



Supplemental Type Certificate

Number: SVR 527

This certificate issued to: AMT Helicopters Pty Ltd
Unit 6/19 Lear Jet Drive
Caboolture, QLD 4510

certifies that the change in the type design for the following product when installed in accordance with the limitations and conditions specified herein meets the airworthiness requirements listed in the continuation sheet.

Original Product:	Type Acceptance Certificate Number:	R61
	Referenced TC Number (if applicable):	EASA.R.140
	Make:	Agusta Westland S.p.A
	Model:	AB206A, AB206B

Description of Type Design Change: Installation of Tail Rotor Blades Part No. AMT-206-1 in accordance with Installation and Maintenance Manual CTRB-IMM-AB206, Revision IR or later approved revision.

Limitations and Conditions:

Prior to incorporating this design change, the installer shall establish that the interrelationship between this change and any other modifications incorporated into the aircraft will not adversely affect the airworthiness of the modified aircraft.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate shall remain in effect until suspended, cancelled or a termination date is otherwise established by the Civil Aviation Safety Authority.

Date of application: 28 July 2015
Date of issuance: 02 November 2015

This certificate is issued pursuant to Regulation 21.113A of the Civil Aviation Safety Regulations.


Peter Nikolic
Delegate of the Authority

This certificate may be transferred in accordance with CAR (1998) 21.47.

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(Continuation Sheet)

Limitations

The following limitations are supplementary to the conditions and limitations given in the original product reference certificate. The original conditions and limitations continue to apply unless superseded by the limitations below.

Applicability

This STC is limited to Composite Tail Rotor Blades Part No. AMT-206-1 manufactured in accordance with a current CASA manufacturing approval.

Airworthiness Limitations

The Composite Tail Rotor Blade is limited to 5000 hours time-in-service (TIS).

Type Design Data

1. Master Records Index - Doc No. 215/552/MRI, Revision IR, dated 30 September 2015, or later CASA approved revision.
2. Flight Manual Supplement - Doc. No. 215/552/FMS, Revision IR, dated 30 September 2015 or later CASA approved revision.
3. Installation and Maintenance Manual – Doc No. CTRB-IMM-AB206, Revision IR, dated 30 September 2015 or later CASA approved revision.

Certification Basis for the components and areas affected:

CAR 6 effective 20 December 1956:

6.111, 6.112, 6.113, 6.114, 6.115, 6.120, 6.121, 6.122, 6.123, 6.130, 6.140, 6.225

and

FAR 27 effective 1 February 1965 including amendments up to FAR 27-44:

27.1, 27.177, 27.301, 27.303, 27.305, 27.307, 27.391, 27.395, 27.397, 27.571, 27.601, 27.602, 27.603, 27.605, 27.607, 27.609, 27.610, 27.611, 27.613, 27.619, 27.623, 27.625, 27.629, 27.653, 27.659, 27.661, 27.671, 27.1301, 27.1309, 27.1529, 27.1565, 27.1581