



Supplemental Type Certificate

Number: SVA 524

This certificate issued to:

Dasyl Avionics Pty Ltd
83a Evelyn Street
Macquarie Fields NSW Australia 2564

certifies that the change in the type design for the following product when installed in accordance with the limitations and conditions as specified herein meets the airworthiness requirements of FAR 25.

Original Product:

Type or Type Acceptance Certificate Numbers:
Referenced TC Number (if applicable):
Make:
Model:

A222 & A108
FAA A16WE
Boeing
737-200/300/400 series

Description of Type Design Change:

Installation of Universal Avionics UNS-1K+ Flight Management System in B737-200 series in accordance with Dasyl Avionics Pty Ltd Installation and Maintenance Manual No. 25-2041-95754 Rev. 0 dated 22 September 2010 or later CASA approved revision.

Installation of Universal Avionics UNS-1Lw Flight Management System (Single or Dual) in B737-300/400 series in accordance with Dasyl Avionics Pty Ltd Installation and Maintenance Manual No. 25-2041-95754-0-2 Rev. 0 dated 23 December 2013 or later CASA approved revision.

Installation of Universal Avionics Monitor/s for ADSB operations (and provide for LP/LPV Approach Option for Single UNS-1Lw Flight Management Systems) in B737-300/400 series in accordance with Dasyl Avionics Pty Ltd Installation and Maintenance Manual No. 25-2041-95754-0-2 Rev. 1 dated 26th June 2014 or later CASA approved revision.

Limitations and Conditions:

For the UNS-1K+ system CASA approved Flight Manual Supplement 2041-95745-12 Rev. 0 dated 23 September 2010 or later CASA approved revision is required.

For the UNS-1Lw systems, (including Monitors) CASA approved Flight Manual Supplements:-

2041-95745-12-1 Rev. 1-0 dated 5 September 2011 Universal MFD 640 installation,
2041-95745-12-2 Rev. 1 dated 26th June 2014 Universal UNS-1Lw Single/Dual Installation,
2041-95745-12-3 Rev. 0 dated 22 January 2011 Universal MFD 640 Installation,
2041-95754-12-4 Rev. 1 dated 26th June 2014 Honeywell TR67A Mode S Transponder Installation,
or later CASA approved supplements as applicable are required.

Prior to incorporating this design change, the installer shall establish that the relationship between this change and any other modifications incorporated into the aircraft will not adversely effect the airworthiness of the aircraft.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that approval.

This certificate and the supporting data, which is the basis for approval, shall remain in effect until suspended, cancelled or a termination date is otherwise established by the Civil Aviation Safety Authority.

Date of application:
Date of issuance:

13th January 2010
22nd September 2010

Date reissued:
Date amended: 22 July 2014

This certificate is issued pursuant to Part 21.113A of the Civil Aviation Safety Regulations


Peter Nikolic
Delegate of the Civil Aviation Safety Authority

This certificate may be transferred in accordance with CASR 21.047.