

AIRWORTHINESS BULLETIN

AWB 32-027 Issue 1 – 13 February 2018

Cessna 441 Main Landing Gear Trunnion Fatigue Cracking

1. Effectivity

All Cessna 441 series aircraft with Main Landing Gear Trunnion P/N 5741115 type or similar fitted.

2. Purpose

This AWB is issued to provide information and recommendations relating to fatigue cracking of main landing gear trunnions.

3. Background

A previous ATSB report investigated a runway excursion incident on landing that resulted from a fracture failure of a P/N 5741115-13 left hand main landing gear trunnion.

A second recent incident has occurred on an overseas aircraft with an identical fracture failure of a P/N 5741115 type main landing gear trunnion.

Laboratory evaluation of both failures indicated that pre-existing cracking with associated fatigue crack growth was the likely cause of each failure.

A review of the two failures shows that there is a risk of fatigue failure of main landing gear trunnions.



Figure 1. Cessna 441 Main Landing Gear Configuration Source: ATSB Report AO-2013-072





Figure 2. Failed Left Hand Main Landing Gear Trunnion P/N 5741115-13 Source: ATSB Report AO-2013-072



Figure 3. Trunnion removed from aircraft showing fracture location Source: ATSB Report AO-2013-072



4. Recommendations

CASA strongly recommends that all operators and maintainers of Cessna 441 aircraft with P/N 574115 type or similar main landing gear to carry out the following;

- 1. Ensure main landing gear trunnions are cleaned and carefully inspected per the applicable Supplemental Inspection Document (SID).
- 2. Carry out a targeted surface high frequency eddy current inspection of the trunnion gusset/web region for evidence of cracking concurrently with the aircraft manufacturer's specified periodic inspection.
- 3. If cracks are present, the affected trunnion should be replaced as a fracture failure during landing may cause significant damage to the aircraft and may cause injury to the occupants.

5. Documentation

ATSB Occurrence Investigation <u>AO-2013-072</u>.

6. Enquiries

Enquiries with regard to the content of this Airworthiness Bulletin should be made via the direct link email address:

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Or in writing, to:

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